

Research Article

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Efficient Implementation of an Accurate Algebraic Scheme for Sharp Interface Advection in Multiphase Flows

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Abstract

This study presents an efficient algebraic scheme known as MULES for sharp interface advection, verified against various schemes including first-order upwind, second-order central, van Leer flux limiter, and Geometric Volume-of-Fluid (VOF). Two problems involving a droplet in a two-dimensional (2D) vortex and a stationary droplet were examined. The model assessed the effects of the Interface Compression (IC) coefficient, ranging from 0 to 2, analyzing parameters such as Interface Advection Error (IAE) and Mass Conservation Error (MCE). Results indicated that increasing IC values enhanced interface tracking accuracy but introduced non-physical instabilities at higher values, compromising mass conservation. Specifically, the IAE decreased from 4.8% to 3.95% as IC increased from 0 to 2, showing a favorable effect until IC surpassed 1.4, where IAE fluctuated around 4%. Conversely, the MCE rose steeply from 0% to 23.19%, driven by parasitic currents and numerical instabilities. Additionally, MULES and van Leer flux limiter schemes evaluated volume fraction smoothing effects. Initial filtering reduced Dimensionless Pressure Difference (DPD) and Capillary Number (Ca), stabilizing the solution, but excessive filtering reintroduced numerical errors and instabilities. With one filtering step, DPD reduced by 0.23 and Ca dropped significantly by 73.31%, improving solution stability. However, further filtering increased DPD and Ca, reflecting the reintroduction of numerical errors. The maximum velocity of parasitic flow around the droplet initially decreased by almost 75% but increased by 30.92% with excessive filtering. IAE increased from 0.7 to 0.9 with initial filtering, then decreased to 0.63 with additional steps, indicating improved solver performance on smoother interfaces.

Keywords: Sharp Interface Advection, Multiphase Flows, Numerical Instabilities, Parasitic Currents, Algebraic Scheme

1. Introduction

Two-phase flows, a term commonly used in computational modeling, pertain to issues involving two fluid phases. On the other hand, multiphase flows encompass a broader spectrum, extending to particle-laden flows. Despite analytical studies like Plateau on such flows dating back to the 19th century, the scope of analytical work is often significantly limited, even for relatively straightforward problems [1]. Experimental observations for practical applications pose challenges due to the difficulty of adapting experimental techniques to multiphase flows [2]. Consequently, there arises a necessity for the development of numerical techniques that are not only accurate and cost-effective

but also guarantee physical consistency. These methods should adeptly capture the interface, seamlessly integrating it with the momentum conservation equation. Unfortunately, developing such approaches proves to be complex due to the inherent challenges associated with multiphase flows. These difficulties encompass a range of issues, including but not restricted to Bestion: (1) Upholding the conservation of mass, momentum, and kinetic energy, (2) Modeling variations in properties across interfaces, particularly significant shifts in density and viscosity, (3) Handling intricate topologies and the differentiation of scales, (4) Ensuring stability in simulating multiphase flows, and (5) Precisely incorporating surface tension forces while maintaining

accuracy [3-5].

Figure 1 presents an overview of various categories within twophase modeling, emphasizing prominent methods that currently garner significant attention in the community. While two-fluid models prove effective for addressing simple problems, they often fall short when handling realistic scenarios [2]. In-depth analyses of two-phase flow models are extensively covered in the works of Ishii and Hibiki and Prosperetti & Tryggvason [2,6]. For readers intrigued by Marker-and-Cell (MAC) techniques, comprehensive discussions can be found in the works of McKee comprenents we discussions can be found in the works of MeKee \sim [10-15], respect al. [7]. Similarly, those interested in Front-Tracking (FT)

 $\frac{d}{d}$ [3-5]. by Tryggvason et al. [8,9]. Additionally, the realm of diffusepresents an overview of various categories within two-
interface approaches encompasses Phase Field (PF), Constrained Interpolation Profile (CIP), and Conservative Level-Set (CLS) ignificant attention in the community. While two-fluid methods. In contrast, sharp-interface approaches involve different classes of Interface-Capturing and Interface-Tracking 1 short when handling realistic scenarios [2]. In-depth methods. It is important to highlight certain abbreviations in Figure 1, such as SPH, LBM, LS, and CLSVOF, which represent represent the Smoothed-Particle Hydrodynamics, Lattice Boltzmann (2,6). The Smoothed-Particle Hydrodynamics, Lattice Boltzmann Method, Lattice Boltzmann Method, Lattice Boltzmann (2,7), and Method, and Method, and Method, Method, Level-Set and Coupled Level-Set and Volume-of-Fluid [10-13], respectively.

Figure 1: Categorization of Numerical Techniques for Handling Two-Phase Flows. The Definitions of Abbreviations can be Found in the Main Text

this context, advection is accomplished by redistributing the unlike compressive schemes, these methods eliminate r conceptualized and created [14]. According to Figure 1, there compression factor into the advection equation for On the other hand, algebraic methodologies avoid explicit mitigating interface dispersal resulting from numerical or geometric VOF schemes necessitate intricate geometric prior studies [15,26-28]. This study primarily centers on one-fluid models, methodologies for capturing interfaces, and Volume-of-Fluid (VOF) techniques [4]. In the VOF approach, the interface delineation occurs implicitly through the allocation of volume fractions to represent the presence of a specific fluid within computational cells. In fluid's content among neighboring cells, facilitated by its movement across the faces of the computational mesh. Since the introduction of the original VOF techniques recorded in the literature numerous varied VOF approaches have been exist two overarching classifications within this method: geometric methodologies involve a detailed reconstruction of the interface using data related to volume fractions [12,13]. reconstruction efforts. Algebraic schemes, renowned for their relative simplicity of implementation, enhanced efficiency, and adaptability to unstructured meshes, operate without the constraints of structured meshes. Nevertheless, their foundation relies on heuristic considerations, rendering them less accurate in comparison to their geometric counterparts [15]. Conversely, operations, resulting in a more cumbersome implementation process and slower execution [5]. It is worth noting that the pursuit of geometric VOF methods tailored for unstructured meshes constitutes a dynamic and ongoing area of research [16-22]. Illustrated in Figure 1, algebraic VOF methods are categorized into two groups: 1) THINC and 2) compressive.

THINC schemes, as proposed by Xiao et al. encompass a collection of recently developed techniques within algebraic VOF methods [23]. These methods assume a hyperbolic-tangent profile for the volume fraction within the computational cell

 $\frac{5}{10}$ in $\frac{5}{10}$ interfaces, and Volume-of-Fluid (VOF) techniques As asserted by Xie et al. these approaches exhibit accuracy y through the allocation of volume fractions to represent computational cost [21]. As a result, THINC schemes are nt across the faces of the computational mesh. Since number at the local cell level [24]. The second group is also ic methodologies involve a detailed reconstruction of anegative value. This term functions by constricting the volume simplicity of implementation, enhanced efficiency, proportion of phases [25]. The compressive impact varies based Interface a detailed reconstructed reconstructed reconstruction of the interface using data related to the interface compression term. containing the interface and obtain the fluxes algebraically. levels similar to those of geometric schemes but with a lower increasingly garnering attention in current research. Additionally, unlike compressive schemes, these methods eliminate necessity for artificial compression and are not dependent on the Courant known as Interface Compression (IC) methods, which are highresolution schemes. These formulations integrate an artificial compression factor into the advection equation for volume fraction, leading to an unconventional diffusion coefficient with fraction distribution perpendicular to the fluid interface, thereby mitigating interface dispersal resulting from numerical diffusion. Consequently, it supports the limitations and preservation of the on the mesh resolution as well as intrinsic IC coefficient, which represents a parameter linked to the artificial compression term. resolution and the coefficient of the interaction, as outlined in prior studies [15,26-28].

> Deshpande et al. identified the occurrence of parasitic currents induced by the implementation of the IC technique in flows primarily governed by the effects of interfacial tension [15,29]. Parasitic flows manifest as resilient abstract vortices in close proximity to the interface, arising autonomously without external influences due to inaccuracies in calculating interface curvature or disparities between surface tension and pressure gradient forces. Typically, these issues manifest in simulations of static bubbles and droplets within fluids characterized by high-density ratios [30]. Hoang et al. investigated how the IC coefficient

influences various factors such as highest velocity attained, the thickness of the VOF interface, and the occurrence of parasitic currents [31]. They convincingly demonstrated that the optimal condition for preventing parasitic flows along with numerical diffusion corresponds to an IC coefficient of 1. Furthermore, their research emphasized the importance of cell dimensions in include criteria s determining the optimal conditions for IC coefficient.

In this investigation, our aim is to utilize a precise algebraic approach for resolving the interface advection equation within approach for resolving the interface advection equation within the context of two-phase flow. This method is commonly known 2. Physical as the Multidimensional Universal Limiter for Explicit Solution (MULES) algorithm [15]. Our approach involves implementing this method through Python coding, as it is currently accessible at coordinates ($x_c = 0.5$ m, $y_c = 0.75$) within the OpenFOAM software. While this open-source software provides users with convenient utilization of this method, it has yet to be made available as an in-house code. The rest of the article is organized as follows: Firstly, the problem statement and modeling approach are introduced in sections 2.1 to 2.3. Subsequently, section 2.4 provides details on the numerical solution procedure, including the discretization of equations. The verification follows in section 3, where results from various schemes, such as first-order upwind second-order central van Leer flux limiter, and geometric VOF, are compared.

Section 4.1 explores the impact of changing the IC coefficient is of the VOF interface, and the occurrence of parasitic on the method's accuracy $[9,32-34]$. In section $[4.2$, the value of surface tension is determined using the results of interface n for preventing parasitic flows along with numerical capturing with two methods, MULES and van Leer flux limiter (as verification), and its accuracy is assessed. Sections β and β include criteria such as Interface Advection Error (IAE), Mass Conservation Error (MCE), Elapsed Time (ET), Dimensionless Pressure Difference (DPD), and Capillary number (Ca). The article concludes with a summary of major findings in section 5.

2. Physical and Mathematical Modelling 2.1 Problem Statement

In Figure 2, a bubble with a diameter $D_0=0.3$ *m* is positioned at coordinates ($x_c = 0.5$ m, $y_c = 0.75$ m) within a square domain measuring 1 m in length (*l*=1 *m*). A two-dimensional velocity provides users with convenient utilization of this field, featuring a periodicity of $T = 2$ s, is applied to induce a $\frac{1}{T}$ it has yet to be made available as an in-house code. The two-dimensional (2D) vortex. Within this dynamic velocity field, the droplet undergoes a reciprocating rotational motion it and modeling approach are introduced in sections throughout a 2-second interval [35]. The volume fraction of the droplet is denoted as α and is defined as 1, while its immediate surroundings exhibit a volume fraction of the discretization of surroundings exhibit a volume fraction of 0. Also, the interface also the interface s. The verification follows in section 3, where results of the droplet is precisely characterized by a volume fraction of 0.5.

Figure 2: The Computational Domain; Regions Colored in Green and Blue are the Surroundings and Droplet, Respectively.

2.2 Governing Equations

 $\partial \alpha_q$

In multiphase flow modeling using the VOF method as the one- all phases is equal to 1, the volume fraction of the pi fluid formulation [9], the volume fraction advection equation is solved according to equation (1), along with the continuity and solving the equation f momentum equations [12]. Typically, this equation is solved for

the secondary phase. Since the sum of the volume fractions of nodeling using the VOF method as the one- all phases is equal to 1, the volume fraction of the primary phase fluid can be obtained by a simple subtraction operation after solving the equation for the volume fraction of the secondary phase, as shown in equation (2).

$$
\frac{\partial a_q}{\partial t} + \mathbf{\nabla} \cdot (\mathbf{U} \alpha_q) = 0 \tag{1}
$$
\n
$$
\alpha_p = 1 - \alpha_q \tag{2}
$$

to the primary (surroundings) and secondary (droplet) phases, it determines the curvature of the interface, surf symbols. For example, U is the velocity vector field created by involving phases with high density ratios, even where α is the volume fraction, and the subscripts *p* and *q* refer respectively. Also, tensorial quantities are indicated by bold 2D vortex.

oundings) and secondary (droplet) phases, it determines the curvature of the interface, surface tension, le, U is the velocity vector field created by involving phases with high density ratios, even small errors The accurate calculation of the volume fraction is crucial because and pressure gradients. However, in multiphase flow problems in the calculation and distribution of the volume fraction can

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cause significant and severe changes in the effective properties. originates from using the velocity field derive Additionally, since the phase interface is confined to only a verage of the velocities of all phases. This weight From the simulation of the simulation's dependency on grid defined according to equation (3) under the simulation's dependency on grid defined according to equation (3) under the size increases within this range when using the VOF method. influence of each phase's speed on interface change Rusche [36] introduced a convective term to equation (1), which phase's speed on interface changes is directly related to the volume fraction of that phase [37]. phase's speed on interface changes is directly related to the volume fraction of that phase [37].

d severe changes in the effective properties. originates from using the velocity field derived from the weighted the phase interface is confined to only a average of the velocities of all phases. This weighted average is defined according to equation (3) under the assumption that the equation (3) under the assumption that the defined according to equation (3) under the assumption that the defined according to equation (3) under the assump In this range when using the VOF method. influence of each phase's speed on interface changes is directly related to the volume fraction of that phase [37].

$$
\mathbf{U} = \alpha_q \mathbf{U}_q + (1 - \alpha_q) \mathbf{U}_p \tag{3}
$$

To integrate equations (1) and (3), the relative velocity, U_r , between the two phases is defined as follows [38]:

$$
\boldsymbol{U}_r = \boldsymbol{U}_p - \boldsymbol{U}_q \tag{4}
$$

 \mathcal{L} (\mathcal{L}) and \mathcal{L} are the set of \mathcal{L} (\mathcal{L}). MULES algorithm can be expressed as follows $[37]$: Therefore, by combining equations (1), (3), and (4), the advection equation that governs the volume fraction of the fluid used in the MII ES also with a separate as follows [37]. \mathcal{L} (a) \mathcal{L} (b) \mathcal{L}

$$
\frac{\partial \alpha_q}{\partial t} + \nabla \cdot \left(\mathbf{U} \alpha_q \right) + \nabla \cdot \left[\mathbf{U}_r \alpha_q \left(1 - \alpha_q \right) \right] = 0 \tag{5}
$$

meaning but is used to compress the phase interface during the be calculated based on the volume fraction gradien the fluid volume fraction, i.e., 0 and 1, and only operates at the Surface Force (CSF) method to model it [40]. In interface. Consequently, it more accurately tracks the interface tension force per unit volume of the flu The third term in this equation does not have a specific physical meaning but is used to compute \mathcal{L} The third term in this equation does not have a specific physical fraction equation has been solved, the surface tens
meaning but is used to compress the phase interface during the she calculated based on the volume fracti and reduces numerical errors resulting from numerical diffusion. calculated using the following equation: Notably, this term equals zero at the lower and upper limits of phases [39]. Brackbill et al. proposed using the $\frac{1}{\sqrt{1}}$ numerical solution of the volume fraction conservation equation. arises due to the excess pressure gradient at the inte and reduces numerical errors resulting from numerical diffusion. Calculated using the following equation.
Now that the problem of the numerical solution of the volume interface. Consequently, it more accurately tracks the interface Where are numerical subdiviry of the volume

s equation does not have a specific physical fraction equation has been solved, the surface tension force can
to compress the phase interface during the she calculated based on the volume fraction gradient. This force f the volume fraction conservation equation. arises due to the excess pressure gradient at the interface between explans zero at the lower and upper initials of phases [39]. Brackom et al. proposed using the community estimation, i.e., 0 and 1, and only operates at the Surface Force (CSF) method to model it [40]. In this method, ntly to compress the phase interface during the be calculated based on the volume fraction gradient. This force quals zero at the lower and upper limits of phases [39]. Brackbill et al. proposed using the Continuum the surface tension force per unit volume of the fluid element is calculated using the following equation:

$$
\boldsymbol{F}_{\sigma} = \sigma \kappa \nabla \alpha_q \tag{6}
$$

 ϵ_{σ}
Where σ and κ represent the surface tension coefficient and the mean radius of interface curvature, respectively. The mean radius of
interface curvature is defined as follows: interface curvature is defined as follows: Where σ and κ represent the surface tension coefficient and the mean radius of interface curvature, respectively. The mean radius of e is defined as follows:
 rv e is defined as follows: $\ddot{}$ s fc ${\rm w}$ s: $\sqrt{7}$

$$
\kappa = -\nabla \cdot \left(\frac{\nabla \alpha_q}{|\nabla \alpha_q|}\right) \tag{7}
$$

numerical solution of the volume fraction equation has been solved, the surface tension force can $S_{\rm eff}$ method to model it. In this method, the surface tension for the surface tension for \sim $S_{\rm eff}$ method to model it. In this method, the surface tension for the surface tension for \sim \mathcal{O} solution of the volume fraction equation for surface tension for the surface tension force can be expected, the surface can be expected, the surface tension for \mathcal{O} It should be noted that the given flow velocity components in relation to the stream function, ψ , must be [41]:

$$
u = \frac{\partial \psi}{\partial y} \tag{8}
$$

$$
v = -\frac{\partial \psi}{\partial x} \tag{9}
$$

Where ψ can be defined as follows [35,42]: Where ψ can be defined as follows [35,42]: Where ψ can be defined as follows [35,42]:

$$
\psi(x, y, t) = \frac{1}{\pi} \cos\left(\frac{\pi t}{T}\right) \sin^2\left(\frac{\pi x}{l}\right) \sin^2\left(\frac{\pi y}{l}\right)
$$
\n(10)

Therefore, there is no need to solve the continuity and momentum equations to obtain velocity field. need to sol e the conti Therefore, there is no need to solve the continuity and momentum equations to obtain velocity field. o need to solve the co ntinuity an<mark>c</mark> momentur Therefore, there is no need to solve the continuity and momentum equations to obtain velocity field.

field. field. field. field. **2.3 Dimensionless Parameters**

2.3 Dimensionless Parameters
The objective parameters described in the previous sections can be alternatively expressed as [5,17,43,44]:

$$
IAE = \sum_{i,j} (\alpha_{i,j}^T - \alpha_{i,j}^0) dxdy
$$
\n
$$
MCE = \frac{|V^T - V^0|}{V^0} = \frac{|\sum_{i,j} \alpha_{i,j}^T dxdy - \sum_{i,j} \alpha_{i,j}^0 dxdy|}{\sum_{i,j} \alpha_{i,j}^0 dxdy}
$$
\n
$$
ET = t^T - t^0
$$
\n
$$
DPD = \frac{D_0 \Delta p}{2\sigma}
$$
\n
$$
Ca = \frac{\mu_q U_{max}}{\sigma}
$$
\n(15)

state, and the final state, respectively. Additionally, *V* denotes the and $0.1 \frac{N}{m}$, respectively. Based on this information, the dimensionless parameters of the dimensionless parameters of the dimensionless paramete state, respectively. Additionally, *V* denotes the and 0.1 N/m , respectively. Based on this information, three In these parameters, the subscripts, j and the superscripts of 0 and *T* represent the 2D computational cell in the domain, the initial volume of the droplet, μ represents the dynamic viscosity, and U_{max} indicates the maximum velocity of the parasitic current. In ne subscripts, *j* and the superscripts of 0 and this study, the density ratio, $\frac{\mu_q}{\rho_p}$ the dynamic viscosity ratio, $\frac{\mu_q}{\mu_p}$. the domain, the initial and the surface tension coefficient are considered to be 20, 1, epresents the dynamic viscosity, and governing (design) dimensionless parameters can be defined as

A $\frac{3}{2}$ $\frac{(\Delta \rho D_0^3}{\mu_q^2} = 1.03 \times 10^4$ 2 0^4 (16) \times 10⁴ (10) (17)

follows [5]:

$$
\text{Eo} = \frac{\Delta \rho g D_0^2}{\sigma} = 171\tag{17}
$$
\n
$$
\text{Mo} = \frac{\text{Eo}^3}{\text{Ar}^4} = 4.44 \times 10^{-10}\tag{18}
$$

numbers, respectively. Furthermore, equation (14) is derived The schematic of computational grid is shown in l dimensional domain. Therefore, the value of the DPD should centers of the cells, while the velocity component be equal to 1 in the ideal mode, making it a reliable criterion for their faces. Interpolation is used to obtain the des
evaluating different schemes in section β 2.1411 Where Ar, Eo, Mo, and are Archimedes, Eotvos, and Morton 2.4 Numerical Solution evaluating different schemes in section 4.2 [41]. Whenever necessary. using the analytical solution for pressure difference in a two-

2.4 Numerical Solution

vely. Furthermore, equation (14) is derived The schematic of computational grid is shown in Figure 3. The in. Therefore, the value of the DPD should centers of the cells, while the velocity components are stored at e ideal mode, making it a reliable criterion for their faces. Interpolation is used to obtain the desired quantity
the person is used to 1 in the interpretation is used to obtain the desired quantity quantities of pressure and volume fraction are stored at the $f(x)$ are stored at the centers of the centers of the velocity components are stored at the velocity components a $\frac{1}{1}$ used to obtain the desired quantity whenever necessary. whenever necessary.

Figure 3: The Computational Grid; Cells Colored in Orange and Red are the Main and Ghost Cells, Respectively

, and the surface tension coefficient are coefficient are considered to be $20, 1, 2, 1, 3, 1, 2$

Now it is time to perform the volume integration on both sides of equation (5) , which is as follows $[9]$:

$$
\int_{V} \frac{\partial \alpha_q}{\partial t} dV + \int_{V} \mathbf{\nabla} \cdot \left[\mathbf{U} \alpha_q + \mathbf{U}_r \alpha_q \left(1 - \alpha_q \right) \right] dV = 0 \tag{19}
$$

form can be obtained as follows [15]:
form can be obtained as follows [15]: Rv annlying the vergence theore By applying the divergence theorem on the advection term and using the explicit Euler method for the unsteady term, the discretized

$$
\alpha_{q_{i,j}^{n+1}} = \alpha_{q_{i,j}^{n}} - \frac{\Delta t}{V_{i,j}} \sum_{f} F_f^n \tag{20}
$$

Here, the subscript of f and the superscript of n represent the cell, and a cell face and the current time step, respectively. Furthermore, expressed as a summation over the cell faces. The Len face and the current time step, respectively. Furthermore, expressed as a summation over the centrales. The contract the contract of V and F_n are time step size, volume of the computational and odvertive flux can f_{ij} , and f_{f} are three step size, volume of the compatitional and advective flux can be obtained.

(23)

current time step, respectively. Furthermore, expressed as a summation over the cell faces. The cell volume Δt , V_{ij} , and F_f^n are time step size, volume of the computational and advective flux can be obtained as follows: cell, and advective flux on the cell face. The advection term is and advective flux can be obtained as follows:

$$
V_{i,j} = (dxdydz)_{i,j} \stackrel{2D}{\rightarrow} V_{i,j} = (dxdy)_{i,j}
$$
 (21)

$$
F_f^n = F_u^n + \lambda_m F_c^n \tag{22}
$$

where r_u , r_c , and α_m denote the first order apwing
are expressed by [15]: Where F_u^n , F_c^n , and λ_m denote the first-order upwind flux, corrected flux, and MULES limiter, respectively. The fluxes F_u^n and F_c^n $\sum_{i=1}^{n}$ \mathbf{r} are expressed by \mathbf{r} Where F_{n}^{n} , F_{n}^{n} , and λ_{n} denote the first-order upwind flux, corrected f 51^{\degree} $\overline{\mathcal{N}}$ λ_m denote the first-order upwind flux, corrected flux, and MULES limiter, respectively. The λ_m r: $\frac{1}{\sqrt{1-\frac{1$ e λ_m denote the first-order upwind flux, corrected flux, and MULES limiter, respectively. The fluxes F_u^n and F_c
51: $5 \cdot \frac{1}{15}$ W_b \overline{a} \overline{H}

$$
F_u^n = \left(\phi_f \alpha_{q_{f,U}}\right)^n \tag{23}
$$

$$
F_c^n = \left(\phi_f \alpha_{q_f}\right)^n + \left[\phi_{rf} \alpha_{q_{rf}} \left(1 - \alpha_{q_{rf}}\right)\right]^n - F_u^n \tag{24}
$$

 \mathcal{L}_{tr} Where the volume flux, ϕ_f and corrected volume flux, ϕ_{rf} are assigned by: \mathcal{L}_{y} Where the volume flux, ϕ_f and corrected volume flux, ϕ_{f} are assigned by: where the volume hax, φ_f and corrected volume hax, $\varphi_{r\hat{f}}$ are assigned by.

$$
\phi_f = U_f S_f = |U_f| n_f S_f \tag{25}
$$

 \mathcal{L}

$$
\phi_{rf} = U_{rf}.S_f = \min_{f \in V_{i,j}} \left[I C \frac{|\phi_f|}{|S_f|}, \max_{f \in Domain} \left(\frac{|\phi_f|}{|S_f|} \right) \right] n_f.S_f \tag{26}
$$

Infinitum operation is performed focally within each cent. In can be determined by:
addition in equations (25) and (26), H_{eff} and S approach the operation is applied actoss an races in the domain, whereas the surface area vector of the centrace, respectively. These quantities
minimum operation is performed locally within each cell. In can be determined by:
additio addition, in equations (25) and (26), U_f , n_f and S_f represent the It is important to note that in equation (26), the maximum face velocity, the unit normal vector to the inter It is important to note that in equation (26) , the maximum face vector of the surface vector of the surface α $\frac{1}{10}$ is important to note that in equation (20), the maximum accuracy operation is applied across all faces in the domain, whereas the surface

note that in equation (26), the maximum face velocity, the unit normal vector to the interface, and the It is important to the domain, whereas the surface area vector of the cell face, respectively. These quantities is not compatible within each cell. In seen he determined by: can be determined by:

$$
U_f = \frac{U_p + U_N}{2} \tag{27}
$$

$$
\boldsymbol{n}_f = \frac{(\nabla \alpha_q)_f}{\left| (\nabla \alpha_q)_f \right| + \frac{10^{-8}}{3\sqrt{V_{i,j}}}}
$$
(28)

$$
S_f = \begin{cases} dy \ i &; \text{vertical face} \\ dx \ j &; \text{horizontal face} \end{cases} \tag{29}
$$

order to limit interface small interface small μ .

equation (24), can be calculated using the relations below $\mathcal{L}_{\mathcal{A}}$

 $\frac{1}{2}$ ln equation (26) value ranging from 0 to 2 in order to limit interface smearing. below [5,15].
Eurthermore, α in equation (22) along with α and In equation (26), the variable IC is typically treated as a constant $\alpha_{q_{rf}}$ in Furthermore, $\alpha_{q_{f,U}}$ in equation (23) \overline{a} Furthermore, $\alpha_{q_{f,U}}$ in equation (23), along with α_{q_f} and

equation (24), can be calculated using the relations below \mathcal{S}

 $\alpha_{q_{rf}}$ in equation (24), can be calculated using the relations below [5,15].

$$
\alpha_{q_{f,U}} = \begin{cases} \alpha_{q_p} & \text{if } \phi_f \ge 0 \\ \alpha_{q_N} & \text{if } \phi_f < 0 \end{cases} \tag{30}
$$

$$
\alpha_{q_f} = \alpha_{q_p} + \frac{\alpha_{q_N} - \alpha_{q_p}}{2} \left[1 - \xi(\phi_f) (1 - \omega_f) \right]
$$
\n(31)

$$
\alpha_{q_{rf}} = \alpha_{q_p} + \frac{\alpha_{q_p} - \alpha_{q_p}}{2} \left[1 - \xi(\phi_f)(1 - \omega_{rf}) \right]
$$
\n(32)

Where $\zeta(\phi)$, ω , and ω represent the step function, van Leer's flux limiter function and the flux limiter of the IC scheme, respectively [33]. Their definitions are as follows: $\frac{1}{2}$ and represent the step function, vanishing $\frac{1}{2}$ and $\frac{1}{2}$ and $\frac{1}{2}$, and $\frac{1}{2}$ Where $\zeta(\phi_f)$, ω_f and ω_{rf} represent the step function, van Leer's flux limiter function and the flux limiter of the IC scheme, respectively

$$
\xi(\phi_f) = \begin{cases} 1 & \text{if } \phi_f \ge 0 \\ -1 & \text{if } \phi_f < 0 \end{cases} \tag{33}
$$

$$
\omega_f = \omega(r_f) = \left(\frac{r+|r|}{1+|r|}\right)_{r=r_f} \tag{34}
$$

$$
\omega_{rf} = \min \left[\max \left[1 - \max \left[\left(1 - \left(4\alpha_{q_P} \left(1 - \alpha_{q_P} \right) \right) \right)^2, \left(1 - \left(4\alpha_{q_N} \left(1 - \alpha_{q_N} \right) \right) \right)^2 \right], 0 \right], 1 \right]
$$
(35)

 $\sum_{i=1}^{\infty}$ as follows: In equation (34), r_t denotes the ratio of successive gradients within the computational domain (refer to Figure 3) and can be defined In equation (54), r_f denotes the ratio or successive gradient.
as follows: In equation (34), r_f denotes the ratio of successive gradients within the computational domain (refer to Figure 3) and can be defined

$$
r_f = \frac{\alpha_{q_U} - \alpha_{q_{UU}}}{\alpha_{q_D} - \alpha_{q_U}}\tag{36}
$$

In equations above, the subscription, s, or , and correspond to the current cell, in the correct cell, neighbor cell, first upwind cell, second understanding, a row of these cells is illustrated in In equations above, the subscripts *P, N, U, UU*, and *D* correspond

upwind cell, and first downwind cell, respectively. To enhance In each of the subscripts \mathcal{L}_{in} , \mathcal{L}_{in} , \mathcal{L}_{in} , and the subscripts \mathcal{L}_{in} and \math

Figure 4: The Arrangement of Important Computing Cells Including Current (P), Neighbor (N), First Upwind (U), Second Upwind (U II) and First Downwind (D) Cells (*UU*), and First Downwind (*D*) Cells

zero elsewhere. To compute this quantity, it is necessary to algorithm and is equal to one in the transition region while being below (refer to Figure 5) [45,46]: \mathcal{L} (), second up wind (), and first downwind (), and first downwind (), and first downwind (), and \mathcal{L} Returning to equation (22), λ_m is incorporated in the MULES

determine the local extrema of the volume fraction, as described below (refer to Figure 5) [45,46]:

$$
\alpha_q^{min} = \min\left(\alpha_{q_p}^n, \alpha_{q_N}^n\right) \stackrel{\text{2D}}{\rightarrow} \alpha_q^{min} = \min\left(\alpha_{q_{i,j}^n}^n, \alpha_{q_{i+1,j}^n}^n, \alpha_{q_{i-1,j}^n}^n, \alpha_{q_{i,j+1}^n}^n, \alpha_{q_{i,j-1}^n}^n\right)
$$
\n(37)

$$
\alpha_q^{max} = \max\left(\alpha_{q_p}^n, \alpha_{q_N}^n\right) \stackrel{\text{2D}}{\rightarrow} \alpha_{q_{i,j}^{max}} = \max\left(\alpha_{q_{i,j}^n}, \alpha_{q_{i+1,j}^n}, \alpha_{q_{i-1,j}^n}, \alpha_{q_{i,j+1}^n}, \alpha_{q_{i,j-1}^n}\right)
$$
(38)

Figure 5: The Schematic for Calculating the Local Extrema of the Volume Fraction; Current and Neighbor Cells

To adhere to the lower and upper limits of the volume fraction and to minimize numerical error, it is essential to adjust equations (37) and (38) as follows: $\frac{1}{\sqrt{2}}$ is equations (37) and (38) and (38) and (38) as follows:

$$
\alpha_q^{min,c} = \max(\alpha_q^{min}, 0)
$$
\n
$$
\alpha_q^{max,c} = \min(\alpha_q^{max}, 1)
$$
\n(39)

(A) inflows and outflows of anti-diffusive fluxes from each face of the cell, denoted as A_f^{\dagger} , along with their cumulative sum for inflows and outflows of anti-diffusive fluxes from each face of relations. In A_f^{\dagger} , the negative and positive supersor Subsequently, as shown in Figure 6, the magnitudes of the

 $\mathcal{L}(\mathcal{L}(\mathcal{L}))$ (39) $\mathcal{L}(\mathcal{L}(\mathcal{L}))$ (39) $\mathcal{L}(\mathcal{L}(\mathcal{L}))$ (39) $\mathcal{L}(\mathcal{L}(\mathcal{L}))$ ws of anti-diffusive fluxes from each face of relations. In A_f^{\dagger} , the negative and positive superscripts indicate each cell, P^{\pm} , must be calculated according to the following inflows and outflows, respectively [46,47].

Figure 1 Cumulative Sum for Each Cell **Figure 6:** The Schematic for Calculating the Inflows and Outflows of Anti-Diffusive Fluxes from each Cell Face, Along with their

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$$
A_{i,j+0.5}^{+} = v_{i,j+0.5}^{n} \left[\frac{\alpha_{i,j+1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j}^{n} \right] dx \qquad ; \text{ if } v_{i,j+0.5}^{n} \ge 0
$$
\n
$$
A_{i,j+0.5}^{+} = v_{i,j+0.5}^{n} \left[\frac{\alpha_{i,j+1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j+1}^{n} \right] dx \qquad ; \text{ if } v_{i,j+0.5}^{n} < 0
$$
\n
$$
A_{i,j-0.5}^{+} = v_{i,j-0.5}^{n} \left[\frac{\alpha_{i,j-1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j}^{n} \right] dx \qquad ; \text{ if } v_{i,j-0.5}^{n} < 0
$$
\n
$$
A_{f}^{\mp} = (F_{h}^{n} - F_{u}^{n})^{\mp} \xrightarrow{2D} \begin{cases} A_{i,j-0.5}^{+} = v_{i,j-0.5}^{n} \left[\frac{\alpha_{i,j-1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j}^{n} \right] dx & ; \text{ if } v_{i,j-0.5}^{n} \ge 0 \\ A_{i+0.5,j}^{+} = u_{i+0.5,j}^{n} \left[\frac{\alpha_{i,j+1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j}^{n} \right] dy & ; \text{ if } u_{i+0.5,j}^{n} \ge 0 \\ A_{i+0.5,j}^{-} = u_{i+0.5,j}^{n} \left[\frac{\alpha_{i,j+1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j}^{n} \right] dy & ; \text{ if } u_{i+0.5,j}^{n} < 0 \\ A_{i-0.5,j}^{+} = u_{i-0.5,j}^{n} \left[\frac{\alpha_{i,j+1}^{n} + \alpha_{i,j}^{n}}{2} - \alpha_{i,j}^{n} \right] dy & ; \text{ if } u_{i-0.5,j}^{n} < 0 \\ A_{i-0.5,j}^{-} = u_{i-0.5,j}^{n} \left[\
$$

$$
P^{\pm} = \mp \sum_{f} A_{f}^{\mp} \stackrel{\text{2D}}{\rightarrow} \begin{cases} P_{i,j}^{\mp} = -\left[A_{i,j+0.5}^{\mp} + A_{i,j-0.5}^{\mp} + A_{i+0.5,j}^{\mp} + A_{i-0.5,j}^{\mp}\right] \\ P_{i,j}^{\mp} = \left[A_{i,j+0.5}^{\mp} + A_{i,j-0.5}^{\mp} + A_{i+0.5,j}^{\mp} + A_{i-0.5,j}^{\mp}\right] \end{cases} (42)
$$

where P_h represents the central-direction in the state. The line focal extrema of the volume fraction, along which
essential to calculate the total transported flux, which arises from the first-order upwind flux across Where F_h^n represents the central-differencing flux. Next, it is the local extrema of the volume fraction, along wi the differences in the volume fraction of each cell in relation to

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nts the central-differencing flux. Next, it is the local extrema of the volume fraction, along with the sum of ransported flux, which arises from the first-order upwind flux across all faces, as follows [5,46]:

$$
Q_{i,j}^{+} = \frac{V_{i,j}}{\Delta t} \left(\alpha_q^{max,c} - \alpha_q^{n}_{i,j} \right) + \sum_f F_{u,f}^{n}
$$
\n(43)

$$
Q_{i,j} = \frac{V_{i,j}}{\Delta t} \left(\alpha_q \binom{n}{i,j} - \alpha_q \binom{min,c}{j} - \sum_f F_{u,f}^n \right)
$$
(44)

By introducing an internal loop denoted by v, the MULES limiter can be determined for all faces and the center of the cells as $B_{\rm eff}$ introducing an internal loop denoted by , the MULES limiter can be determined for all faces α follows $[5, 46]$: $\text{follows } [5, 46]:$

$$
\lambda_{m,i,j}^{\mp,\nu+1,n} = \max\left[\min\left(\frac{\pm\sum_{f} \lambda_{m,f}^{\nu,n} A_{f}^{\pm} + Q_{i,j}^{\pm}}{P_{i,j}^{\pm}}, 1\right), 0\right]
$$
\n(45)

$$
\lambda_{m,f}^{\nu+1,n} = \begin{cases} \min(\lambda_{m,P}^{1+\nu+1,n}, \lambda_{m,N}^{-\nu+1,n}) & \text{if } A_f \ge 0\\ \min(\lambda_{m,P}^{-\nu+1,n}, \lambda_{m,N}^{1+\nu+1,n}) & \text{if } A_f < 0 \end{cases} \tag{46}
$$

In this iterative process, the initial value of MULES limiter implementing the MULES algorithm, a numerical solution interface normal vector is directly redelta function approximation is employed to model the standard gradient, along with the surface tension, will be stored must be developed for equations (6) and (7). In this study, the developed function approximation approximation must be developed for equations (6) and (7). In this study, the delta function approximation is v reaches either 2 or 3. After discretizing equation (5) and at the cell faces. Additionally, according to equation must be developed for equations (6) and (7). In this study, the the volume fraction. Therefore, it is also anticipa Letta function approximation is emproyed to moder the standard gradient, along with the continuum Surface Force (CSF) [4,9]. As illustrated in Figure faces. The approximation

MULES algorithm, a numerical solution interface normal vector is directly related to the gradient of In this nerative process, the mittal value of MOLES limiter the mean radius of curvature at the interface is stored at the for all faces is set to $1 (\lambda_{m,f}^{\text{rel},n} = 1)$. This loop continues until cell center, while the un) 7, the mean radius of curvature at the interface is stored at the oximation is employed to model the standard gradient, along with the surface tension, will be stored at the cell cell center, while the unit normal vector to the interface is stored or 3. After discretizing equation (5) and at the cell faces. Additionally, according to equation (28) , the for equations (6) and (7) . In this study, the the volume fraction. Therefore, it is also anticipated that this faces. The approximated form of the surface tension is given by:

$$
\mathbf{F}_{\sigma} = \mathbf{f}_{\sigma} \delta_{s} = -\sigma \kappa \nabla \alpha_{q} \sim -\sigma \kappa \nabla_{h} \alpha_{q}
$$
\n⁽⁴⁷⁾

Here ∇_h denotes the discrete gradient operator. For clarity, equation (47) is discretized for the right face at $(i + 0.5, j)$ as follows [5]:

$$
\boldsymbol{F}_{\sigma,x,i+0.5,j} = -\sigma \kappa_{i+0.5,j} \left(\frac{\partial \alpha_q}{\partial x}\right)_{i+0.5,j} = -\sigma \left(\frac{\kappa_{i,j}^n + \kappa_{i+1,j}^n}{2}\right) \left(\frac{\alpha_q_{i+1,j}^n - \alpha_q_{i,j}^n}{dx}\right)
$$
(48)

Figure *1*. The schematic for Calculating the Mean Radius of Curvature at the interface and the Associated surface fell **Figure 7:** The Schematic for Calculating the Mean Radius of Curvature at the Interface and the Associated Surface Tension $T_{\rm eff}$ mean radius of curvature at the interface, denoted as the interface, denoted as the interface, denoted as $T_{\rm eff}$

The mean radius of curvature at the interface, denoted as $\kappa_{ij}^{\ n}$, can be determined usi radius of curvature at the interfa The mean radius of curvature at the interface, denoted as $\kappa_{i,j}^n$, can be determined using the following equation:
 $\kappa_{i,j}$

$$
\kappa_{i,j}^{n} = -\left[\frac{1}{dx}\left(\eta_{x,i+0.5,j}^{n} - \eta_{x,i-0.5,j}^{n}\right) + \frac{1}{dy}\left(\eta_{y,i,j+0.5}^{n} - \eta_{y,i,j-0.5}^{n}\right)\right]
$$
(49)

Here, $\eta_x^{\ n}$ represents the di $\frac{1}{\sqrt{2}}$ $\overline{}$ Here, $\eta_x^{\ n}$ represents the discretized form of the unit normal vector to the interface, which is expressed as follows:

$$
\eta_{x,i+0.5,j}^{n} = \frac{m_{x,i+0.5,j}^{n}}{\sqrt{\left(m_{x,i+0.5,j}^{n}\right)^{2} + \left(m_{y,i+0.5,j}^{n}\right)^{2} + \frac{10^{-8}}{3\sqrt{V_{i,j}}}}}
$$
(50)

In equation (50), m_x^n and m_y^n represent the component i. In equation (50), m_x^n and m_y^n represent the components of the volume fraction gradient, defined as follows:

$$
m_{y,i+0.5,j}^{n} = 0.25[m_{y,i,j+0.5}^{n} + m_{y,i+1,j+0.5}^{n} + m_{y,i,j-0.5}^{n} + m_{y,i+1,j-0.5}^{n}](51)
$$

$$
m_{y,i,j+0.5}^{n} = \frac{\alpha_{q_{i,j+1}}^{n} - \alpha_{q_{i,j}}^{n}}{dy}
$$
 (52)

$$
m_{x,i+0.5,j}^{n} = \frac{\alpha_{q_{i+1,j}}^{n} - \alpha_{q_{i,j}}^{n}}{dx}
$$
\n(53)

artant to note tension is its dependence on volume fraction. Smoothing this multiple steps. յլ
1 It is important to note that these calculations are straightforward for the other faces as well. A key aspect of calculating surface study $[5,9]$. As shown in Table 1, this process is im $\mathbf{I}^{\mathbf{m}}$ $\frac{1}{\sigma}$ It is input to the theories that the straightforward for the other faces as well. And the other faces as w ension is its dependence on volume fraction. Sinooning this inditione steps.
quantity through filtering may enhance accuracy. This filtering

method can also effectively impact critical parameters in this as well. A key aspect of calculating surface study [5,9]. As shown in Table 1, this process is implemented in multiple steps.

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exactly match those of the first-order upwind solver. Table 2 presents three parameters including **Table 1: The Process of Smoothing the Volume Fraction, Along with its Corresponding Steps and Equations**

2.5 Verification

first-order upwind, second-order central, van Leer flux limiter, methods. a spherical droplet in a 2D vortex ($N_x = N_y = 32$ and $\Delta t = 0.005$ s), inst-order upwind, second-order central, van Leer hux limiter, methods.
and geometric VOF methods. First, it is important to note that if In this section, the verification of the MULES algorithm using as detailed in section 2.1, is examined. The results, including IAE, MCE, and ET, are compared with those obtained from the

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al droplet in a 2D vortex ($N_r = N_r = 32$ and $\Delta t = 0.005$ s), order upwind solver. Table 2 presents three parameters including E, and ET, are compared with those obtained from the time (in seconds) for both the MULES and first-order upwind $\frac{dE}{dt}$ and $\frac{dE}{dt}$ and $\frac{dE}{dt}$ and $\frac{dE}{dt}$ are $\frac{dE}{dt}$ in section, the verification of the MULES algorithm using of the MULES solver should exactly match those of the firstthe MULES limiter, λ , in Equation (22) is set to 0, the results interface advection error, mass conservation error, and elapsed methods.

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Table 2: The Comparison Between the two Interface Capturing Strategies, MULES and First-Order Upwind, Based on **Three Criteria: IAE, MCE, and ET**

 difference in elapsed time (approximately 40 seconds) between the two methods is due to the more complex coding of the and van Leer flux limit MULES solver and its use of different functions, which require these methods are prove more time to achieve the same results as the first-order upwind and usphays the volume fraction method. According to Amani [5], the MULES method has a three selected times $(0, T)$ As observed in Table 2, for two different values of IC in the MULES scheme, the results for the specified parameters are consistent with those of the first-order upwind method, confirming the accuracy of the MULES solver. The slight more time to achieve the same results as the first-order upwind

ved in Table 2, for two different values of IC in the lower computational cost than the geometric VOF method, but the specified parameters the geometric VOF method offers higher accuracy. It is also ng the accuracy of the MULES solver. The slight convergence and stability, and performs more accurately than and van Eeer nux ininter. To demonstrate this, the results for these methods are provided in Table 3. Additionally, Figure 8a-e istent with those of the first-order upwind method, important to ensure that the MULES method achieves good other methods such as first-order upwind, second-order central, and van Leer flux limiter. To demonstrate this, the results for achieve the same results as the first-order upwind displays the volume fraction contours for various schemes at method has a three selected times $(0, T / 2, T)$.

19 **Table 3: The Comparison between the Different Interface Capturing Schemes, Based on Three Criteria: IAE, MCE, and ET**

As shown in Table 3, the computational cost of the MULES method is higher than that of the other methods (approximately 40 seconds). This is expected due to the more complex calculations and formulations involved in MULES. While the accuracy of the MULES method is lower than that of the geometric method, it is on par with, or even better than, the other methods. This is confirmed by comparing the percentages of IAE and MCE. For example, the IAE in the first-order upwind method is nearly twice that of the MULES, second-order central, and van Leer flux limiter methods. Consequently, this method is likely to

n in Table 3, the computational cost of the MULES experience significant numerical diffusion, leading to inaccurate . This is expected due to the more complex calculations achieve more consistent interface tracking. Additionally, the MCE Lower than the contribution of the geometric method, vOF methods is zero, ensuring high stability and preventing Example 1. The other methods is zero, ensuring high stability and preventing Fract with, or even better than, the other methods. This interface holse due to mass differences. In contrast, the MCE for med by comparing the percentages of IAE and MCE. the second-order central difference method is 10.6 nple, the IAE in the first-order upwind method is nearly 13 times and 1.89 times higher than the MULES method with ter methods. Consequently, this method is likely to error indicates that the second-order central difference method interface tracking (see Figure 8c), whereas the other methods for the first-order upwind, van Leer flux limiter, and geometric interface noise due to mass differences. In contrast, the MCE for two different IC values of 0.1 and 0.5, respectively. This higher

is prone to fluctuations at the phase interface, resulting in lower stability (see Figure 8d). Therefore, both the MULES (see Figure ac 8a-b) (Multimedia available online) and van Leer flux limiter

to fluctuations at the phase interface, resulting in lower (see Figure 8e) schemes are stable numerical approaches that accurately track the phase interface.

Figure 8: The volume fraction contours at t=0 (left), t=T/2 (center), and t=T (right) for a) MULES with IC=0.1 (Multimedia available online)

- b) MULES with IC=0.5 (Multimedia available online)
- c) first-order upwind
- d) second-order central
- e) van Leer flux limiter methods. The interface is depicted with a black iso-line of $\alpha=0.5$.

3. Results and Discussion

3.1 Effect of Interface Compression (IC)

As mentioned in section 2.4, the IC value is typically fixed between 0 and 2. However, Berberovic et al. and Deshpande et al. expanded the range and examined this constant value up to 4 [15,48]. If the IC value is zero, no compression will occur at the interface. A value between 0 and 1 indicates moderate compression, while values greater than 1 indicate enhanced

ts and Discussion

compression. In this section, IC values between 0 and 2 are ext of Interface Compression (IC) considered to evaluate their impact on interface dynamics and mass conservation within the droplet (secondary phase). This 0 and 2. However, Berberovic et al. and Deshpande et investigation focuses on a spherical droplet within a 2D vortex problem, using a grid resolution of $N_x = N_y = 32$ and a time step α . If the IC value is zero, no compression will occur of Δt =0.005s, as detailed in section 2.1. Figure 9a-i illustrates the volume fraction contours derived from the MULES method with varying IC values at the final time, *T*.

Figure 9. The volume fraction contours at ℓ T for MULES method with a) iC ℓ , ℓ **Figure 9:** The volume fraction contours at $t=T$ for MULES method with a) IC=0, b) IC=0.25, c) IC=0.5, d) IC=0.75, e) IC=1, f)

thickness becomes more significant, leading to more accurate steeply until the IC reaches and it is substantiate this chain, the results of it is the interface. The interface in the IT for an embedded in Table 4 and Figure 10. According to the data in around 150 seconds. Table 4 and Figure 10, the IAE decreases from 4.8% to 3.95% Figure 9 demonstrates that as the IC value increases, the interface interface tracking. For instance, in Figure 9a, when the IC value is zero, there is no phase within the droplet, and the maximum volume fraction inside the droplet is 0.9. However, as the IC value increases, the maximum volume fraction inside the droplet reaches 1, and the phase inside the droplet is considered in the simulation. As the IC value increases, the compressibility of the interface also increases, leading to the volume fraction of 1 inside the droplet advancing towards the interface. However, if the IC value exceeds 1, non-physical instabilities may form during the simulation, disturbing the mass conservation of the droplet. Therefore, it is recommended to keep the IC value between 0 and 1. To substantiate this claim, the results of IAE and MCE

 $\mathcal{L} = \mathcal{L} \times \mathcal{L}$, and i) , and i) , and i) . The individual set of $\mathcal{L} = \mathcal{L} \times \mathcal{L}$ fraction inside the droplet is 0.9. However, as the IC effect on interface tracking, but this effect diminishes once the s ; the maximum volume fraction inside the droplet IC surpasses the critical value of 1.4. Conversely, the MCE rises on. As the IC value increases, the compressibility of the increase in MCE is due to parasitic currents at the interface let advancing towards the interface. However, if the IC errors in Figure 10, it is evident that IAE is almost independent let ceeds 1, non-physical instabilities may form during of IC changes, while IC significantly affects the droplet mass e, it is recommended to keep the IC value between 0 indicate that changes in IC have no effect on the elapsed time for as the IC value increases from 0 to 2. This decline continues steeply until the IC reaches 1.4, where the IAE reaches 3.8%, after which this error fluctuates around 4%. Therefore, it can be concluded that increasing the IC value initially has a favorable from 0% to 23.19% as the IC increases from 0 to 2. This steep and instability in the numerical solution. By comparing both conservation. Additionally, the time results reported in Table 4 each simulation. The ET for all simulations remains consistently around 150 seconds.

Parameter		Interface capturing method: MULES						
	$IC = 0.01$	$IC = 0.05$	$IC = 0.1$	$IC = 0.2$	$IC = 0.4$			
IAE $(%)$	4.8	4.72	4.63	4.48	4.24			
MCE(%)	0.07	0.36	0.82	1.79	3.71			
ET(s)	153.98	147.96	146.51	147.78	148.77			
	$IC = 0.5$	$IC = 0.7$	$IC = 0.9$	$IC = 1$	$IC = 1.2$			
IAE $(%)$	4.22	4.08	3.97	3.92	3.85			
MCE (%)	5.64	8.01	10.21	11.11	13.24			
ET(s)	149.82	154.78	153.26	152.48	153.75			
	$IC = 1.4$	$IC = 1.5$	$IC = 1.6$	$IC = 1.8$	$IC = 2$			
IAE $(%)$	3.8	$\overline{4}$	4.09	4.04	3.95			
MCE(%)	15.37	18.95	21.37	22.96	23.19			
ET(s)	152.69	152.67	153.53	158.09	151.88			

Table 4: The Comparison Between the Different Values of IC in MULES Scheme, Based on Three Criteria: IAE, MCE, and ET.

Figure 10: The Percentage of Interface Advection Error (IAE) and Mass Conservation Error (MCE) versus the Interface Compression If the IAE exclusively. **Figure 20** The set Provides a Detailed View of the IAE Exclusively.

In this section, surface tension is calculated using the relationships introduced in section 2.4. The

3.2 Effect of Smoothing the Volume Fraction

In this section, surface tension is calculated using the 11a-b present various problem-
In this section into develop a cation 2.4. The integer of column L_{eve} from the idea of $(x_c = y_c = 0.5 \text{ m and } u = v = 0 \text{ m/s})$, using a grid resolution of N_x time, $t = 0.5 \text{ s}$. examined using the filters outlined in Table 1 (C1 to C4). This time. Additionally, Fig relationships introduced in section 2.4. The impact of volume fraction smoothing and noise reduction on this quantity is investigation focuses on a stationary spherical droplet problem

d using the filters outlined in Table 1 (C1 to C4). This time. Additionally, Figure 12a-e displays the volume fraction $\frac{1}{\pi}$ of Smoothing the Volume Fraction $N_y = 32$ and a time step of ∆t=0.00125s. Table 5 and Figure 11a-b present various parameters for both the MULES and van Leer flux limiter methods, including interface advection error, dimensionless pressure difference, capillary number, and elapsed contours along with parasitic current for various schemes at final time, $t = 0.5$ s.

Parameter	Interface capturing method: MULES							
	without filter	C ₁	C ₂	C ₃	C ₄			
IAE $(%)$	0.7	0.9	0.79	0.7	0.63			
DPD	1.22957	0.99209	1.01283	1.01614	1.01509			
Ca	0.02484	0.00663	0.00671	0.00752	0.00868			
ET(s)	136.56	143.74	145.82	147.61	150.06			
Parameter	Interface capturing method: van Leer flux limiter							
	without filter	C ₁	C ₂	C ₃	C4			
IAE $(%)$	0.69	0.9	0.79	0.7	0.63			
DPD	1.22155	0.99208	1.01281	1.01620	1.01509			
Ca	0.02472	0.00663	0.00671	0.00753	0.00868			
ET(s)	130.01	132.21	134.90	138.46	144.18			

Four Criteria: IAE, DPD, Ca, and ET. (C1 to C4 are introduced in Table 1). Table 5: The Comparison Between the two Interface Capturing Strategies, MULES and Van Leer Flux Limiter, Based on

Figure 11: The Results of two Interface Capturing Methods, MULES and The Van Leer Flux Limiter, include: a) DPD and Ca, and b) ET and IAE Versus Number of Filtering.

According to the data in Table 5 and Figure 11a-b, both the MULES and van Leer flux limiter methods perform similarly, indicating that volume fraction smoothing has a consistent effect on both schemes of tracking the interface (see Figure 12a-e) (Multimedia available online). A closer examination of Table 5 and Figure 11a reveals that, as the number of filtering steps increases from 0 to 4, DPD and Ca initially decrease. However, after the first filtering, the values of these two quantities increase. With one filtering step, the interface fluctuations are reduced, leading to a decrease in the maximum velocity of the parasitic flow around the droplet by almost 75%, as shown in Figure 12b (Multimedia available online) and Figure 13. Consequently, the capillary number decreases significantly from 0.02484 to its lowest value of 0.00663, representing a 73.31% reduction. A smaller capillary number, closer to zero, indicates higher stability of the solution. Moreover, with one filtering step, DPD reduces by 0.23, approaching the analytical DPD value of 1. However, with further increases in the number of filtering steps, DPD and CA increase, likely due to numerical errors from excessive smoothing of the volume fraction (see Figure 12c-e) (Multimedia available online). Consequently, DPD tends toward

an approximate limit value of 1.016, experiencing increases of 0.0162 and 0.01509 compared to the analytical DPD value in the third and fourth filters, respectively. Additionally, according to Figure 13, the maximum parasitic velocity around the droplet increases by 30.92% percent from 0.0663 m/s value to 0.0868 m/s, causing an upward trend in CE from 0.00663 to 0.00868 (also refer to Figure 12c-e) (Multimedia available online). Furthermore, according to Table 5 and Figure 11b, filtering the volume fraction once increases IAE from 0.7 to 0.9. This is likely because minimizing parasitic currents and approaching the real pressure difference inside and outside the droplet makes tracking the interface more challenging, thus increasing the numerical error for both methods. However, with an increased number of filtering steps, IAE decreases from 0.9 to 0.63. As more filtering is performed, fluctuations in the volume fraction are further reduced, resulting in smoother interfaces that the solver can track with less error. It should be noted that filtering imposes additional computational cost, approximately 10 to 15 seconds for the MULES method and 2 to 14 seconds for the van Leer flux limiter method, on the simulation.

varying numbers of filtering: a) zero, b) one, c) two, d) three and e) four. The interface is depicted with a black iso-line of α =0.5. f ux available online) schemes with variance α **Figure 12:** The volume fraction contours at final time, $t=0.5$ s, for MULES (left) and van Leer flux limiter (right) schemes with (Multimedia available online)

Figure 13 The maximum velocity of parasitic flow versus number of filtering. **Figure 13:** The Maximum Velocity of Parasitic Flow Versus Number of Filtering.

4. Conclusion

Leer flux limiter, and Geometric VOF. Two problems involving In the present work in the present was the effects of the Interface Compression (IC) coefficient, pitfalls of selecting an excessively high IC value. Addi **In the present work, an embreme algorithment scheme known as**
MULES was implemented for sharp interface advection. This a spherical dioplet in a 2D vottex and a stationary dioplet were in it. This disparity ingifigures the complex relationship investigated. The verified model was then employed to examine the IC value and mass conservation, varying from 0 to 2, representing no compression, moderate MULES and van Leer flux limiter schemes were empl parameters, including Interface Advection Error (IAE), Mass filtering reduces the Dimensionless Pressure Difference In the present work, an efficient algebraic scheme known as method was rigorously verified and compared with several other schemes, including first-order upwind, second-order central, van a spherical droplet in a 2D vortex and a stationary droplet were compression, and enhanced compression. Several key objective Conservation Error (MCE), and Elapsed Time (ET), were analyzed. As the IC value increases, the thickness of the interface becomes more distinct, enhancing the accuracy of the interface tracking within the simulation. At very low IC values, the droplet lacks a discernible phase, and the volume fraction remains limited. However, as the IC value rises, the simulation effectively captures the internal phase of the droplet, leading to a more realistic representation. Interestingly, while the accuracy of interface tracking improves with increasing IC values, this enhancement comes with a caveat. When the IC value surpasses a certain threshold, the simulations begin to exhibit non-physical

EXECUTE: This the simulation, particularly in the form of parasitic currents that the simulation of \mathbb{R}^d . sion, and enhanced compression. Several key objective examine the effects of volume fraction smoothing. Initially, ation Error (MCE), and Elapsed Time (ET), were but beyond the first step, DPD begins to rise again, eventually becomes more distinct, enhancing the accuracy of the This pattern suggests that while some filtering can stabilize the $\frac{1}{2}$ instabilities, which can disrupt the conservation of mass within , including first-order upwind, second-order central, van results. While IAE showed an almost steady improvement as the the droplet. These instabilities manifest as erratic behaviors in arise at the interface, compromising the overall integrity of the IC increased, MCE displayed a heightened sensitivity to changes in IC. This disparity highlights the complex relationship between the IC value and mass conservation, underscoring the potential pitfalls of selecting an excessively high IC value. Additionally, MULES and van Leer flux limiter schemes were employed to filtering reduces the Dimensionless Pressure Difference (DPD), approaching a value slightly above the analytical prediction. system, excessive filtering introduces numerical errors that offset the initial benefits. The Capillary Number (Ca) experiences a significant drop after the first filtering step, indicating improved solution stability. However, additional filtering causes the Ca to increase again, reflecting the same trend observed with DPD. The velocity of parasitic flow around the droplet initially decreases markedly with the first filtering step, leading to reduced interface fluctuations and a more stable system. However, with further

filtering, the parasitic flow velocity starts to increase again, suggesting that excessive filtering can destabilize the system. The IAE behaves differently compared to other parameters. Initially, IAE increases with one filtering step, likely due to the challenge of accurately tracking a more stable interface. However, as more filtering steps are applied, IAE decreases, indicating that the solver can better track the smoother interfaces produced by extensive filtering.

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